

Orientation estimation fusing a downward looking camera and inertial sensors for a hovering UAV

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16th International Conference on Advanced Robotics
School of Engineering
Universidad de la República del Uruguay
November 25-29

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Introduction and objectives

Unmanned Aerial Vehicles (UAV) - Quadrotors

- Advantages: inexpensive, easy to build and to maintain, lightweight and easy to control.
- Disadvantages: very limited payload, and computational resources.

Some applications like inspection or indoor navigation require stationary fly or hovering → **orientation estimation**.



Inertial sensors



Camera



Sonar



Magnetic compass



GPS

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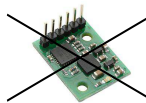
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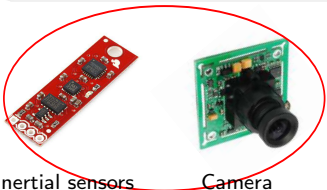
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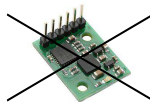


Inertial sensors

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Objectives

Orientation estimation algorithm fusing spectral features based visual yaw angle estimation, and inertial measurements with an EKF (double correction stage).

Visual yaw angle estimation - Rotation estimation

A 3D scene point $M = [X \ Y \ Z \ 1]^T \in \mathbb{P}^3$ is projected to the point $m = [u \ v \ 1]^T \in \mathbb{P}^2$ on the image plane, as (pin-hole model)

$$sm = PM = K[R|t]M$$

where (R, t) relates the WCS and the CCS.

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For a downward looking camera and planar scene ($Z = 0$)

$$sm = K[R|t] \begin{bmatrix} X \\ Y \\ 0 \\ 1 \end{bmatrix} = K \begin{bmatrix} r_1 & r_2 & t \end{bmatrix} \begin{bmatrix} X \\ Y \\ 1 \end{bmatrix},$$

or

$$sm = HM,$$

with

$$H = K \begin{bmatrix} r_1 & r_2 & t \end{bmatrix}.$$

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For a moving camera, at time t_A

$$s_A m_A = H_{WA} M,$$

for the next camera frame, at time t_B

$$s_B m_B \approx H_{WB} M.$$

For smooth motion $s_A \approx s_B$

$$m_A \approx H_{BA} m_B,$$

with $H_{BA} = (H_{WB})^{-1} H_{WA}$ the homography that relates corresponding points $m_A \leftrightarrow m_B$.

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Neglecting the roll and pitch angle

$$m_A = H_{BA} m_B \approx \begin{bmatrix} R_z & t \\ \mathbf{0} & 1 \end{bmatrix} m_B$$

→ Euclidean transformation.

Visual yaw angle estimation - Spectral features

Phase Correlation Method

Given two images i_A and i_B differing only in a displacement (u, v) , such as

$$i_A(x, y) = i_B(x - u, y - v),$$

their Fourier transform are related by

$$I_a(\omega_x, \omega_y) = e^{-j(u\omega_x + v\omega_y)} I_b(\omega_x, \omega_y).$$

The cross power spectrum (CPS) is defined as

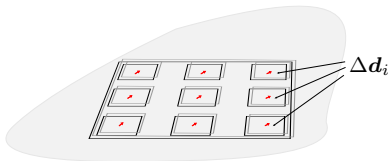
$$\frac{F(\omega_x, \omega_y) G^*(\omega_x, \omega_y)}{|F(\omega_x, \omega_y)| |G^*(\omega_x, \omega_y)|},$$

where G^* is the complex conjugate of G .

$$\begin{aligned} Q(\omega_x, \omega_y) &= \frac{I_a(\omega_x, \omega_y) I_b^*(\omega_x, \omega_y)}{|I_a(\omega_x, \omega_y)| |I_b^*(\omega_x, \omega_y)|} \\ &= e^{-j(u\omega_x + v\omega_y)}, \end{aligned}$$

where $Q(\omega_x, \omega_y)$ is the correlation phase matrix, and the inverse transform is an impulse located in (u, v)

$$\begin{aligned} \mathcal{F}^{-1}[Q(\omega_x, \omega_y)] &= q(x, y) \\ &= \delta(x - u, y - v). \end{aligned}$$



Orientation estimation - Orientation representation

The orientation estimation is performed using the EKF and quaternion orientation representation $\mathbf{q} = [q_0 \ q_1 \ q_2 \ q_3]^T$ with $\|\mathbf{q}\| = 1$.

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Given the angular velocity vector $\boldsymbol{\omega} = [\omega_x \ \omega_y \ \omega_z]^T$, changes in orientation can be expressed by

$$\dot{\mathbf{q}} = \frac{1}{2} \mathbf{q} \times \begin{bmatrix} 0 \\ \boldsymbol{\omega} \end{bmatrix},$$

which represents the time derivative of the orientation quaternion, also written as

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where $\boldsymbol{\Omega}(\boldsymbol{\omega})$ is the skew-symmetric matrix associated to the vector $\boldsymbol{\omega}$, such that

$$\dot{\mathbf{q}} = \frac{1}{2} \begin{bmatrix} 0 & -\omega_x & -\omega_y & -\omega_z \\ \omega_x & 0 & \omega_z & -\omega_y \\ \omega_y & -\omega_z & 0 & \omega_x \\ \omega_z & \omega_y & -\omega_x & 0 \end{bmatrix} \mathbf{q}.$$

Orientation estimation - Camera/IMU fusion

Double correction stage EKF

- Prediction: gyroscope measurements.
- Correction:
 - ▶ accelerometers measurements,
 - ▶ visual yaw angle estimation.

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Extended Kalman Filter

Prediction:

$$\hat{\mathbf{x}}_k^- = F_{k-1} \hat{\mathbf{x}}_{k-1}$$

$$P_k^- = F_{k-1} P_{k-1} F_{k-1}^T + Q_{k-1}$$

Correction:

$$K_k = P_k^- H_k^T (H_k P_k^- H_k^T + R_k)^{-1}$$

$$\hat{\mathbf{x}}_k = \mathbf{x}_k^- + K_k (\mathbf{z}_k - h_k(\hat{\mathbf{x}}_k^-))$$

$$P_k = (I - K_k H_k) P_k^-$$

- Prediction:

$$F_k = I + \frac{\Delta t}{2} \Omega(\boldsymbol{\omega})$$

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- First correction stage:

$$h_{ak}(\mathbf{q}) = g \begin{bmatrix} 2q_1 q_3 - 2q_2 q_0 \\ 2q_2 q_3 + 2q_1 q_0 \\ q_0^2 - q_1^2 - q_2^2 + q_3^2 \end{bmatrix}$$

with $\mathbf{z}_a = [a_x \ a_y \ a_z]^T$.

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- Second correction stage:

$$\psi = \arctan(\gamma)$$

where
$$\gamma = \frac{2(q_0 q_3 + q_1 q_2)}{q_0^2 + q_1^2 - q_2^2 - q_3^2},$$

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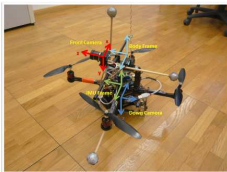
$$\text{with } \mathbf{z}_h = [\cos \psi \ \sin \psi]^T,$$

and

$$h_{h_k}(\mathbf{q}) = \begin{bmatrix} \frac{1}{\sqrt{\gamma^2 + 1}} \\ \frac{\gamma}{\sqrt{\gamma^2 + 1}} \end{bmatrix}$$

Implementation and results

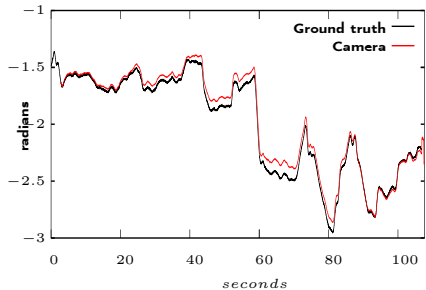
- OpenCV library for image processing and computer vision algorithm.
- MAV dataset of the sFly project ¹, containing
 - ▶ Image sequence obtained by a forward and a downward looking camera.
 - ▶ Measurements from an inertial Measurement Unit (IMU).
 - ▶ Ground truth information given by a Vicon system.



¹<http://www.sfly.org>

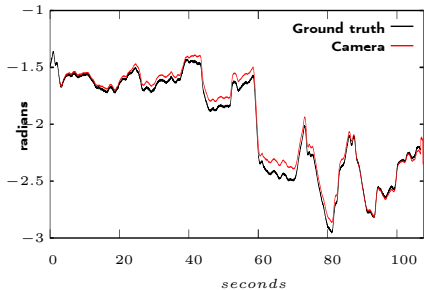
Implementation and results - Visual yaw angle estimation

Estimation for a short fly

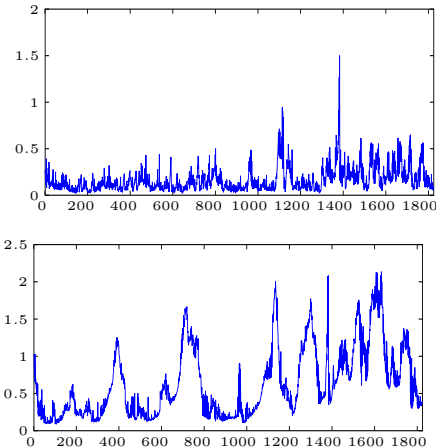


Implementation and results - Visual yaw angle estimation

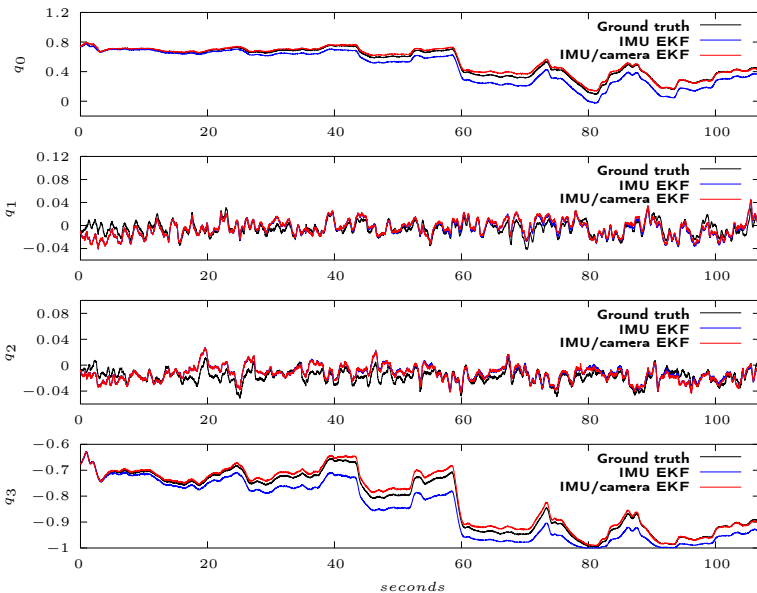
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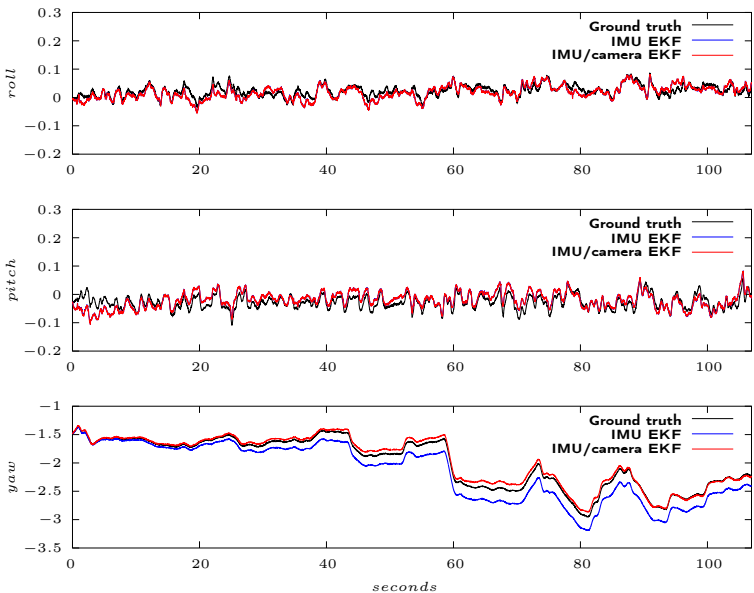
Mean accumulated error ε_{10} and ε_{100} (degrees vs. frame num.)



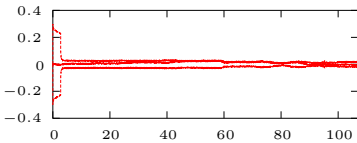
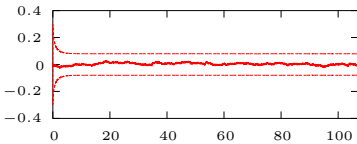
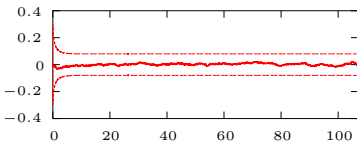
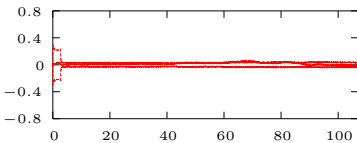
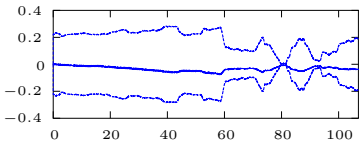
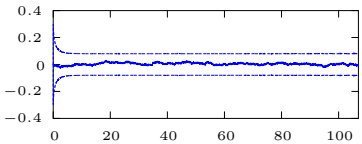
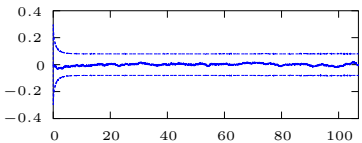
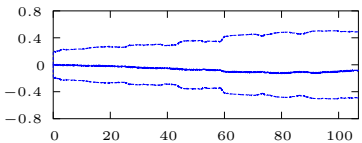
Implementation and results - Estimated quaternion



Implementation and results - Estimated Euler angles



Implementation and results - IMU only vs. Camera/IMU



Conclusion and future work

- New approach for quadrotor orientation estimation fusing inertial measurements with a downward looking camera.
 - ▶ Inertial measurements are mainly for roll and pitch angles estimation,
 - ▶ and yaw angle is estimated by the camera using spectral features.
- Measurement fusing is based on a double correction stage EKF.
- Experimental results have been obtained using a public dataset of a hovering UAV.
- Even though the visual yaw angle estimation has the typically accumulated error, it can be used to reduce the IMU drift.
- Camera-IMU orientation fusion presents a significant reduction in both the bias and drift compared with the IMU only orientation estimation.
- Future work includes the estimation of the position (pose), and the implementation in a real setup.

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Thanks for your attention.